

Lessons Learned, A Project Manager's Perspective



Bryan Fafaul

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WIRE Case Study / Glory Experience





Do NOT Launch on March 4th

WIRE
March 4, 1999



Glory
March 4, 2011





WIRE MRR

January 28, 1999



Mission Description

- The Wide-field Infrared Explorer (WIRE) will be used to conduct a deep infrared, extra galactic science survey 500 times more sensitive than the Infrared Astronomy Satellite (IRAS) Faint Source Catalog. The WIRE instrument is provided by the Jet Propulsion Laboratory and teaming partner, the Space Dynamics Laboratory of Utah State University. The instrument consists of a cryogenically-cooled, 30 cm telescope which will detect faint astronomical sources in two infrared wavelength bands. The WIRE spacecraft bus is provided by the GSFC and is based on the SMEX SWAS spacecraft bus design. The spacecraft bus utilizes an all composite structure and the first of the SMEX•Lite modular solar arrays as part of the Explorer Project's technology infusion program. The WIRE spacecraft will be placed into a 540 km Sun-synchronous orbit using OSC's Pegasus XL launch vehicle in late February, 1999



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SCIENCE TEAM

- Nick Gautier, JPL
- Paul Graf, BASG
- Terry Herter, Cornell
- Jim Houck, Cornell
- Carol Lonsdale, IPAC
- Harvey Moseley, GSFC
- David Shupe, IPAC
- Tom Soifer, Caltech
- Gordon Stacey, Cornell
- Mike Werner, JPL

WIDE-FIELD INFRARED EXPLORER

- Perry Hacking, JSE, Principal Investigator
- Tom Luchik, JPL, Project Manager
- Carol Lonsdale, IPAC, Deputy Project Manager
- JET PROPULSION LABORATORY

GSFC/SMEX WIRE TEAM

- Jim Watzin, Project Manager
- Bryan Fafaul, Mission Manager
- Dave Everett, Spacecraft System Engineer
- Leroy Sparr, Instrument Systems Engineer/COTR

SCIENCE AND ENGINEERING STAFF

- Science Operation
- Data Analysis

- David Henderson, JPL
- Tim Conrow, IPAC
- R L h IPAC
- Olga Pevunova, IPAC
- Fan Fang, NRC
- Cong Xu, IPAC

- Virginia Yoshioka, ADMINISTRATION
- Walt Boyd, RESOURCES
- Teresa Alfery, PROCUREMENT
- Steve Giacoma, COST ANALYSIS

ENGINEERING ASSURANCE

- Bob Axsom, PRODUCT ASSURANCE
- Darrell Schmit, SYSTEM SAFETY

WIRE INSTRUMENT

- Harry Ames, Instrument Program Manager
- John Kemp, Instrument Engineer
- UTAH STATE UNIV/SPACE DYNAMICS LABORATORY

- C/SSR
- PERFORMANCE MEASUREMENT

- Jon Buttars
- Rosemarie Jorgensen

PRODUCT ASSURANCE

- Richard Austin

FOCAL PLANE ARRAYS & ASSOCIATED ELECTRONICS

- John Kemp, SDL

CRYOGEN SYSTEM

- Scott Schick, SDL

OPTICAL SYSTEM

- Duane Miles, SDL

ELECTRONICS

- Wally Gibbons, SDL

CRYOGENIC OPERATIONS

- Scott Schick, SDL

INSTRUMENT ENGINEERING

- John Kemp, SDL

BOEING NORTH AMERICAN

- Bill Armstrong

LOCKHEED MARTIN ADVANCED TECH CENTER

- Brenda Costanzo

MECHANICAL SYSTEMS

- Mehrdad Roosta, SDL

Instrument Development Organization



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Mission Historical Summary

- SMEX Mission Set #2 selections announced 9/93
- Competitive Definition Phase followed, resulting in WIRE being selected as the fifth SMEX mission in 12/94
- Development Phase started in 10/95 after a one year mission phasing delay (cost saving measure)
- Spacecraft bus integration began 10/97 with the delivery of the flight qualified primary structure and was completed in 1/98 with the integration of the flight instrument electronics
- Cryostat assembly underwent separate environmental test program at LMATC/SDL prior to delivery to GSFC. High fidelity thermal and mass simulators were used with spacecraft prior to cryostat delivery
- Cryostat arrived at GSFC 5/98, was integrated with the spacecraft, and completed remaining observatory environmental testing



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Mission Historical Summary (continued)

- The WIRE spacecraft successfully conducted environmental test program (3/98-6/98)
 - EMI/EMC testing
 - Vibration testing
 - Thermal balance/thermal vacuum testing
 - Acoustics testing (with cryostat)
 - Magnetic calibration (with cryostat)
 - Comprehensive performance testing (with cryostat)
 - Numerous L&EO and nominal science mission operations simulations
- Extensive functional testing after each environmental test has been performed to insure proper function of the spacecraft



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Mission Historical Summary (continued)

- Mission followed traditional GSFC review cycle with additional instr. reviews
 - Detailed instrument reviews -- conducted by JPL
 - FPA and Electronics 2/28/96
 - Science 3/1/96
 - SO&DA 3/4/96
 - Cryostat 3/6-7/96
 - Safety 3/8/96
 - Electronics 3/12/96
 - Optics 3/12/96
 - Instrument I&T 3/13/96
 - Instrument Single Design Review 3/19-20/96
 - Instrument Pre-Environmental Review 5/29/97
 - Instrument Pre-Ship Review 5/6/98
- Mission reviews -- conducted by GSFC
 - Single Design Review 5/21-23/96
 - Pre-Environmental Review 2/23/98
 - Flight Operations Readiness Review 7/21/98
 - Pre-Ship Review 7/22/98



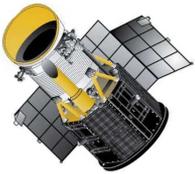
WIRE MRR

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Observatory Status

- All observatory electronics now have accumulated ~1500 hours of trouble-free testing
- There are NO open anomalies against observatory, GSE, or ground system
- There are NO critical safety verifications open
- All launch site safety documentation and procedures are signed-off by all parties, including the range
- Spacecraft and GSE shipped on January 15, 1999, arrived at VAFB, California January 18, 1999, using standard SMEX practices
 - Air ride, environmental controlled tractor trailer
 - Shock mounted, enclosed, instrumented, N2 purged shipping container



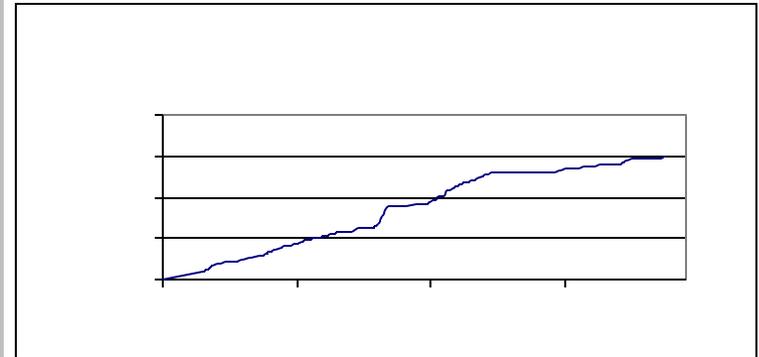
WIRE MRR

January 28, 1999



Operating Hours as of January 22, 1999

Component	Total Hours Prior to PSR	Total Hours Since PSR	Total Hours
SPE/Shunt Box	1100	384	1484
SCS	1100	384	1484
ACE	861	321	1182
Receiver	650	384	1034
Transmitter	300	209	509
A Wheel	580	226	806
B Wheel	592	227	819
C Wheel	511	170	681
Y Wheel	829	298	1127
Gyro 1	416	97	513
Gyro 2	422	175	597
Gyro 3	465	231	696
WIE	450	244	694
Pyro Box	120	5	125
Star Tracker	100	5	105
WAES	550	224	774





Internal / External Lessons Learned



There are six key lessons learned from the WIRE failure:

1. Proper application of FPGAs
2. Proper Peer Reviews
3. Effective closed-loop tracking of actions
4. Managing across organizational boundaries
5. Extra vigilance required when deviating from full system end-to-end testing
6. System design must consider both nominal and off nominal scenarios

GAO

United States General Accounting Office

Report to the Subcommittee on Space and Aeronautics, Committee on Science, House of Representatives

January 2002

NASA

Better Mechanisms Needed for Sharing Lessons Learned

Table 1: Reasons for Spacecraft Failures

Reasons for Failure	Major Program Reviews			Major Mishap Reviews							
	Broad Area Review	Lockheed Martin Independent Assessment Team	Faster, Better, Cheaper Task Force	W ^a I R E	Mars Climate Orbiter	Mars Polar Lander	Lewis	S ^b O H O	Mars Observer	DC-X ^c	Challenger
Cost and Schedule Constraints	•			•			•		•		•
Insufficient Risk Assessment and Planning		•	•		•	•	•	•	•	•	•
Underestimation of Complexity and Technology Maturity	•		•				•	•	•		
Insufficient Testing			•	•	•		•		•	•	•
Poor Team Communication	•	•	•	•	•	•	•	•			•
Inattention to Quality and Safety	•	•	•		•						•
Inadequate Review Process	•	•	•	•	•	•	•	•		•	•
Design Errors	•			•	•	•			•	•	•
Inadequate System Engineering	•	•	•	•	•	•	•	•	•		
Inadequate or Under Trained Staff	•	•			•	•			•	•	•



GAO – Lessons Learned Report

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Inattention to Quality and Safety	•	•	•		•						•
Inadequate Review Process	•	•	•	•	•	•	•	•		•	•
Design Errors	•			•	•	•			•	•	•
Inadequate System Engineering	•	•	•	•	•	•	•	•	•		
Inadequate or Under Trained Staff	•	•			•	•		•	•	•	•



Common Findings / Observations

- Cost / Schedule Constraints
- Insufficient Testing
- Poor Team Communication
- Inadequate Review Process
- Design Errors
- Inadequate Systems Engineering



Summary

- The Scientist needs to get involved and stay engaged
- Everyone is a Systems Engineer like it or not
- Success / Protection comes in layers
- Listen to every opinion, especially the tough ones
- Make the good technical decisions first
- Understand programmatic trades / risks
- Recognize “Management Anomalies” and do something
- Don’t be afraid to speak-up
- Be proactive, make a positive difference